### **European Aviation Safety Agency**



# SUPPLEMENTAL TYPE CERTIFICATE

#### EASA.A.S.03767

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 1702/2003 to

## Scandinavian Avionics Design GmbH

Stratusvej 9 7190 Billund Denmark

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Product Type Certificate Number: FAA TC / TCDS No. A24CE;

EASA TC / TCDS No. EASA.IM.A.277

Type Certificate Holder: Hawker Beechcraft Corporation

Model: 200, 200C, B200, B200C, 300, 300LW,

B300, B300C, 1900C

**Description of Design Change:** 

Installation of Sandel SA4550 EADI and Sandel SA4500 EHSI Collins AHS-1000

### **Associated Technical Documentation:**

- EASA approved Engineering Order, Doc. No. 2050691-EO, issue 1, dated 25-May-2009, or later approved issue

- EASA approved Aircraft Flight Manual Supplement, Doc. No. 2050691-AM, Issue 1, dated 01-Jun-2009 or later approved issue

 Approved Instructions for Continued Airworthiness, Doc. No. 2050691-CA, Issue 1, dated 01-Apr-2009 or later approved issue

#### **Limitations and Conditions:**

 This change is limited to the configuration of aircraft / equipment type as listed under Item 4. in EASA approved Engineering Order, Doc. No. 2050691-EO, issue 1, dated 25-May-2009, or later approved issue.

2. The installation of this modification by third persons is subject to written permission of the approval holder, and having the appropriate approved documentation at one's disposal.

3. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.

This Certificate shall remain valid unless otherwise surrendered or revoked.

For the European Aviation Safety Agency,

Date of issue: 01 July 2009

Manfred REICHEL
Project Certification Manager

 $\mathscr{C}$  – EASA.A.S.03767 – Scandinavian Avionics Design GmbH

EASA Form 91, Issue 1